

1. LQR: Linear Quadratic Regulator

Lecture 15

ECE/ME 179D

• System dynamics are LINEAR

• Cost function is QUADRATIC

• Controller REGULATES states, i.e., drives them to zero (and not to a reference trajectory)

Topics:

1. LQR
2. "rel" versus "abs" (again...)

- LQR approach sets the values in the controller gain matrix, K , for the control law:

$$u = -Kx$$

$(m \times 1) \quad (m \times n)(n \times 1)$

n = # of states
 m = # of inputs, (e.g., torques)

Such that the quadratic cost function below is MINIMIZED:

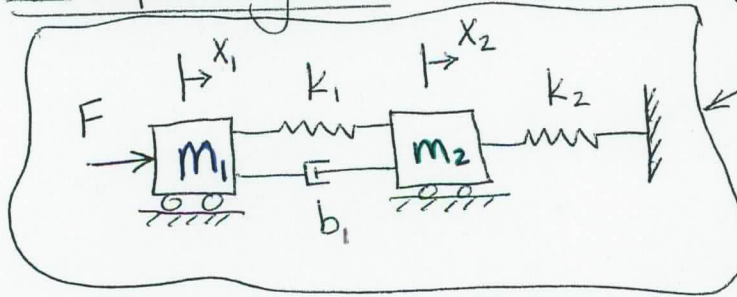
$$J = \int_{t=0}^{\infty} \{ x^T Q x + u^T R u \}$$

$(1 \times n)(n \times n)(n \times 1) \quad (1 \times m)(m \times m)(m \times 1)$

of course, " x " is " $x(t)$ ", " u " is " $u(t)$ ", i.e., they are varying w/ time

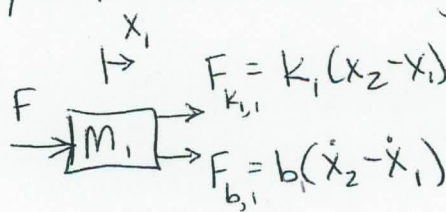
- Both Q & R are symmetric, positive definite.
- Often, Q & R are simply diagonal matrices, in practice.
- When $m=1$ (one actuator), R is 1×1 , so: $u^T R u = R u^2$

Example System: Two-mass "spring-mass-damper"...



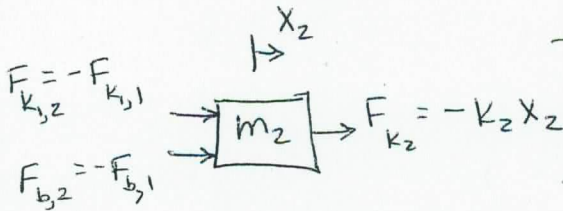
See also MATLAB code

By inspection of a free-body diagram for each mass:



EOM 1

$$m_1 \ddot{x}_1 = F + k_1(x_2 - x_1) + b_1(\dot{x}_2 - \dot{x}_1)$$



EOM 2

$$m_2 \ddot{x}_2 = -k_1(x_2 - x_1) - b_1(\dot{x}_2 - \dot{x}_1) - k_2 x_2$$

Using absolute coordinates for x_1 & x_2 , let's define the state vector, X , as:

$$X = \begin{bmatrix} x_1 \\ x_2 \\ \dot{x}_1 \\ \dot{x}_2 \end{bmatrix}$$

Then the EOMs in matrix form are:

$$\dot{X} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \ddot{x}_1 \\ \ddot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -\frac{k_1}{m_1} & \frac{k_1}{m_1} & -\frac{b_1}{m_1} & \frac{b_1}{m_1} \\ \frac{k_1}{m_2} & -\frac{(k_1+k_2)}{m_2} & \frac{b_1}{m_2} & -\frac{b_1}{m_2} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{m_1} \\ 0 \end{bmatrix} F$$

$\dot{X} =$

A

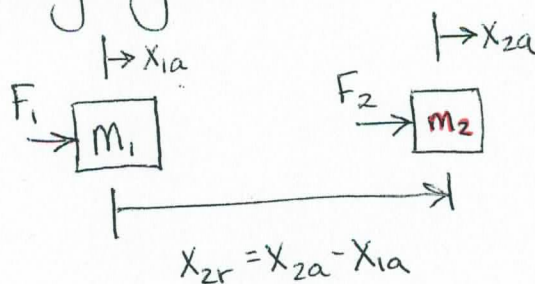
X

$+ B u$

$$\dot{X} = AX + Bu$$

(See MATLAB code for example that creates a state space system & uses LQR for feedback control...)

2. Let's also use a simplified two-mass system to better understand how to determine " $\tilde{\tau}$ " for the Lagrangian EOMs...



← Systems are completely decoupled:

$$\begin{aligned} m_1 \ddot{x}_{1a} &= F_1 \\ m_2 \ddot{x}_{2a} &= F_2 \end{aligned}$$

See also
"L12 Supplement"
(Lecture 12)

• Lagrangian approach:

GC's?	x_{1a} (absolute), x_{2a} (abs)	x_{1a} (abs), x_{2r} (rel)
T^* ?	$T^* = \frac{1}{2} m_1 \dot{x}_{1a}^2 + \frac{1}{2} m_2 \dot{x}_{2a}^2$	$T^* = \frac{1}{2} m_1 \dot{x}_{1a}^2 + \frac{1}{2} m_2 (\dot{x}_{1a} + \dot{x}_{2r})^2$
V ?	$V = 0$	$V = 0$
$\mathcal{L} = T^* - V$	$\mathcal{L} = T^*$	$\mathcal{L} = T^*$

EOM 2: $\frac{d}{dt} \left(\frac{\partial \mathcal{L}}{\partial \dot{x}_{2a}} \right) - \frac{\partial \mathcal{L}}{\partial x_{2a}} = \tilde{\tau}_{2a}$

(2A) $m_2 \ddot{x}_{2a} = F_2$

$\frac{d}{dt} \left(\frac{\partial \mathcal{L}}{\partial \dot{x}_{2r}} \right) - \frac{\partial \mathcal{L}}{\partial x_{2r}} = \tilde{\tau}_{2r}$

(2R) $m_2 (\ddot{x}_{1a} + \ddot{x}_{2r}) = F_2$

LHS's are the same, $\therefore$ RHS's must also match.

EOM 1: $\frac{d}{dt} \left(\frac{\partial \mathcal{L}}{\partial \dot{x}_{1a}} \right) - \frac{\partial \mathcal{L}}{\partial x_{1a}} = \tilde{\tau}_{1a}$

(1A) $m_1 \ddot{x}_{1a} = F_1$

$\frac{d}{dt} \left(\frac{\partial \mathcal{L}}{\partial \dot{x}_{1a}} \right) - \frac{\partial \mathcal{L}}{\partial x_{1a}} = \tilde{\tau}_{1r}$

$m_1 \ddot{x}_{1a} + m_2 (\ddot{x}_{1a} + \ddot{x}_{2r}) = ? = \tilde{\tau}_{1r}$

$\tilde{\tau}_{1r} = \tilde{\tau}_{1a} + \tilde{\tau}_{2a}$
 $= F_1 + F_2$

RHS (1A) RHS (2A)

$(\text{LHS of } (1A)) + (\text{LHS of } (2A)) = (\text{RHS of } (1A) + \text{RHS of } (2A))$ (3)

```

1 clear all
2 % This code models a two-cart system, as describe in class
3 % during Lecture 15. A force is input at m1. m1 and m2
4 % are connected by k1 and b1, and k2 connects m2 to a wall.
5 % Katie Byl, 2012.

```

```

6
7 m1 = 5; m2 = 2;
8 k1 = 200; b1 = 10; k2 = 100;

```

```

9
10 A = [0 0 1 0
11       0 0 0 1
12      -k1/m1 k1/m1 -b1/m1 b1/m1
13      k1/m2 -(k1+k2)/m2 b1/m2 -b1/m2];
14 B = [0 0 1/m1 0]';
15 C = diag([1 1 1 1]);
16 D = 0;

```

```

17
18 % Commands below create a "state space system"

```

```

19 ss1 = ss(A,B,C,D)

```

```

20

```

```

21 % Open loop poles are the eigenvalues of matrix A

```

```

22 open_loop_poles = eig(A)

```

```

23

```

```

24 %=== Code below finds feedback gains "K" via LQR (linear
25 % quadratic regulator) approach from Lecture 15. ===

```

```

26

```

```

27 % Q gives penalties on STATE (squared errors), and

```

```

28 % R gives a penalty on the square of the control effort, u.

```

```

29 % (i.e.,  $J = x^*Qx + u^*Ru$ , from Lecture 15.)

```

```

30 % Usually, Q has more than one non-zero diagonal entry, but

```

```

31 % note it still works fine as shown below. In general, Q will

```

```

32 % be diagonal (with no off-diagonal terms), however:

```

```

33 Q = diag([1 0 0 0]);

```

```

34 R = .001;

```

```

35

```

```

36 % MATLAB can calculate the optimal gain matrix, K, for the

```

```

37 % LQR problem defined by the system dynamics (A and B matrices)

```

```

38 % and your costs, Q and R:

```

```

39 K = lqr(A,B,Q,R);

```

```

40

```

```

41 % Below is the CLOSE-LOOP system, using  $u = -Kx$  as a control law:

```

```

42 ss2 = ss(A-B*K,B,C,D);

```

```

43

```

```

44

```

```

45 % ===== We will then compare the above system to one with a new R =====

```

```

46

```

```

47 Qnew = Q;

```

```

48 % Below, let us change ONLY the value of R (with same Q):

```

```

49 Rnew = 1;

```

```

50 Knew = lqr(A,B,Qnew,Rnew);

```

```

51 % Define a TIME vector, to simulate the response, using MATLAB:

```

```

52 dt = 0.001;

```

```

53 t = 0:dt:50;

```

```

54 % Below, we also define an "initial condition" for the system:

```

```

55 X0 = [0;0;1;0];

```

```

56

```

```

57 % ===== Now, we will do LOTS of plots, both of STATES and of

```

```

58 % control effort, u

```

```

59 figure(11); clf

```

```

60 % "initial" simulated the initial condition response, given X0.

```

```

61 y1 = initial(ss1,X0,t);

```

```

62 for n=1:4

```

```

63     subplot(5,1,n)

```

```

64     plot(t,y1(:,n),'b-');

```

```

65 end

```

```

66 u1 = 0*t; % Free response has "u=0", always...

```

```

67 subplot(515)

```

```

68 plot(t,u1,'r-');

```

```

69 subplot(511)

```

```

70 title('Free Response (u=0)')

```

```

71

```

```

72 J1terms(5) = 0; % u1=0

```

```

73 for n=1:4

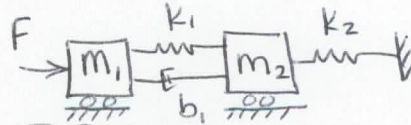
```

```

74     J1terms(n) = dt*sum(y1(:,n).^2);

```

Matlab m-file
"sample_lqr.m"



$\dot{X} = AX + Bu$ ← Defines EOMS.
 $y = CX + Du$ ← Defines measured outputs, y.

Type "help ss" in Matlab for usage.

Open-loop poles.

$Q \dot{=} R$, for LQR.

"help lqr" for usage... Sets "K" gains.

Closed-loop system, w/ additional disturbance input " Bu_{dist} " $u_{tot} = u + u_{dist}$

New R $\dot{=}$ corresponding K_{new}

Start w/ $x_1 = x_2 = 0, \dot{x}_1 = 1, \dot{x}_2 = 0$

initial condition response

free response, $u=0$

$u_{tot} = -Kx + u_{dist}$, to allow step input in u_{dist} closed loop.

```

75 end
76 fprintf('sqrd er: X(1)    X(2)    X(3)    X(4)    u\n')
77 fprintf('u=0: %8.3f %8.3f %8.3f %8.3f %8.3f \n', J1terms)
78
79 input('\n\nNow, let's use LQR feedback control:')
80 figure(12); clf
81 y2 = initial(ss2,X0,t);
82
83 % Closed-loop poles are eigenvalues of "A-BK", for u=-K*x
84 closed_loop_poles1 = eig(A-B*K)
85
86 for n=1:4
87     subplot(5,1,n)
88     plot(t,y2(:,n),'b-');
89     hold on
90 end
91 subplot(515)
92
93 % Line below calculates the CONTROL EFFORT, u:
94 u2 = -y2*K';
95
96 hold on
97 plot(t,u2,'r-'); % u is the actuation, assuming "u=-K*x" is the law
98 J2terms(5) = dt*sum(u2.^2);
99 for n=1:4
100     J2terms(n) = dt*sum(y2(:,n).^2);
101 end
102 fprintf('\n');
103 fprintf('C1: Q1:%5.3f Q2:%5.3f Q3:%5.3f Q4:%5.3f R:%5.3f \n',...
104     Q(1,1),Q(2,2),Q(3,3),Q(4,4),R(1))
105 fprintf('C1: %8.3f %8.3f %8.3f %8.3f %8.3f \n',J2terms)
106 J2add = J2terms.*[diag(Q)', R];
107 fprintf('J1 = %8.3f + %8.3f + %8.3f + %8.3f + %8.3f = %12.8f \n',...
108     J2add, sum(J2add));
109
110
111 % Set to 1 or 0, below, to tweak vs redefine R and Q...
112 if 0
113     % try "tweaking" K:
114     input('Now, let's TWEAK the values in K:')
115     Kfac = .7+.6*rand(1,4)
116     Knew = Kfac.*K;
117     Qnew = Q; Rnew = R;
118 else
119     input('\n\nNow, let's try LQR with same Q but larger R penalty:')
120     Rnew
121 end
122
123 % Closed loop poles with Qnew and Rnew, below:
124 closed_loop_poles2 = eig(A-B*Knew)
125
126 figure(12); %clf
127 ss3 = ss(A-B*Knew,B,C,D);
128 y3 = initial(ss3,X0,t);
129 for n=1:4
130     subplot(5,1,n)
131     plot(t,y3(:,n),'k-');
132     Ax = axis;
133     axis([0 15 Ax(3:4)])
134 end
135 subplot(515)
136 u3 = -y3*Knew';
137 plot(t,u3,'k-');
138 Ax = axis;
139 axis([0 15 Ax(3:4)])
140 subplot(511)
141 title('u = -B*K, with K set via LQR')
142
143 J3terms(5) = dt*sum(u3.^2);
144 for n=1:4
145     J3terms(n) = dt*sum(y3(:,n).^2);
146 end
147 fprintf('\n');
148 fprintf('C2: Q1:%5.3f Q2:%5.3f Q3:%5.3f Q4:%5.3f R:%5.3f \n',...

```

} ← Repeat initial condition response for closed-loop system, using K.

} ← closed-loop poles

} ← $u = -Kx$

} ← u^2 , summed over time

} ← x^2

$$J = \int_0^{\infty} x^T Q x + u^T R u$$

$$J \approx \sum_{n=1}^N (Q(1,1)x_1^2 + Q(2,2)x_2^2 + \dots + Q(3,3)x_3^2 + Q(4,4)x_4^2 + Ru^2) \Delta t$$

(Q is diagonal, & we approximate the continuous-time integral w/ a sum.)

to test cost J when K is "tweaked" around its optimal value

} ← CL poles, using Knew

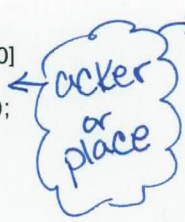
} ← CL system with Knew

} ← $u = -K_{new}x$

```

149 Qnew(1,1),Qnew(2,2),Qnew(3,3),Qnew(4,4),Rnew(1))
150 fprintf('C2: %8.3f %8.3f %8.3f %8.3f %8.3f \n',J3terms)
151 J3add = J3terms.*[diag(Qnew)', Rnew];
152 fprintf('J2 = %8.3f + %8.3f + %8.3f + %8.3f + %8.3f = %12.8f \n',...
153 J3add, sum(J3add));
154
155 % Now, try pole placement
156 pdes = [-20 -30 -100 -200]
157 Kplace = acker(A,B,pdes)
158 ss4 = ss(A-B*Kplace,B,C,D);
159 y4 = initial(ss4,X0,t);
160 u4 = y4*Kplace';
161
162 figure(13); clf
163 for n=1:4
164     subplot(5,1,n)
165     plot(t,y4(:,n),'b-');
166     Ax = axis;
167     axis([0 .5 Ax(3:4)])
168 end
169 subplot(515)
170 plot(t,u4,'r-');
171 Ax = axis;
172 axis([0 .5 Ax(3:4)])
173 subplot(511)
174 title('u = -B*K, with K set via Pole Placement')
175
176 J4terms(5) = dt*sum(u4.^2);
177 for n=1:4
178     J4terms(n) = dt*sum(y4(:,n).^2);
179 end
180 fprintf('\n');
181 fprintf('\n');
182 fprintf('C2: Q1:%5.3f Q2:%5.3f Q3:%5.3f Q4:%5.3f R:%5.3f \n',...
183 Qnew(1,1),Qnew(2,2),Qnew(3,3),Qnew(4,4),Rnew(1))
184 fprintf('C2: %8.3f %8.3f %8.3f %8.3f %8.3f \n',J4terms)
185 J4add = J4terms.*[diag(Qnew)', Rnew];
186 fprintf('Jp = %8.3f + %8.3f + %8.3f + %8.3f + %8.3f = %12.8f \n',...
187 J4add, sum(J4add));
188
189
190

```



pole placement:

K is now set (as needed) to place the 4 poles at (arbitrary), but stable, locations on the s-plane.

(See Workspace output & plots, next pages.)

Lecture 15

ECE/ME 179D

MATLAB Output:

```
1
2 ss1 =
3 a =
4      x1  x2  x3  x4
5      x1   0   0   1   0
6      x2   0   0   0   1
7      x3  -40  40  -2   2
8      x4  100 -150  5  -5
```

$$A = \begin{bmatrix} \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots \end{bmatrix}$$

```
9
10 b =
11      u1
12      x1   0
13      x2   0
14      x3  0.2
15      x4   0
```

$$B = \begin{bmatrix} \vdots \\ \vdots \\ \vdots \\ \vdots \end{bmatrix}$$

```
16
17 c =
18      x1  x2  x3  x4
19      y1   1   0   0   0
20      y2   0   1   0   0
21      y3   0   0   1   0
22      y4   0   0   0   1
```

C defines outputs

```
23
24 d =
25      u1
26      y1   0
27      y2   0
28      y3   0
29      y4   0
```

D is often all zeros.

31 Continuous-time state-space model.

```
32
33 open_loop_poles =
34  -3.4361 + 12.8869i
35  -3.4361 - 12.8869i
36  -0.0639 + 3.3525i
37  -0.0639 - 3.3525i
```

← OL poles

```
38 Q =
39      1   0   0   0
40      0   0   0   0
41      0   0   0   0
42      0   0   0   0
```

for LQR

$$K_1 = [8.8, -2.5, 7.0, 2.0]$$

```
43 R =
44  1.0000e-03
```

```
45 sqrd er: X(1)  X(2)  X(3)  X(4)  u
46 u=0:  0.240  0.126  2.696  1.448  0.000
```

for R=0.001

49 Now, let's use LQR feedback control:

```
50 closed_loop_poles1 =
51  -3.4367 + 12.8855i
52  -3.4367 - 12.8855i
53  -0.7656 + 3.4439i
54  -0.7656 - 3.4439i
```

```
55
56 C1: Q1:1.000 Q2:0.000 Q3:0.000 Q4:0.000 R:0.001
57 C1:  0.018  0.010  0.226  0.145  17.045
58 J1 =  0.018 + 0.000 + 0.000 + 0.000 + 0.017 = 0.03513940
```

Increasing penalty on control effort ($u^T R u$) decreases control gains, typically.

61 Now, let's try LQR with same Q but larger R penalty:

w/R=1.0

```
62 Rnew =
63      1
64 closed_loop_poles2 =
65  -3.4361 + 12.8869i
66  -3.4361 - 12.8869i
67  -0.0685 + 3.3526i
68  -0.0685 - 3.3526i
```

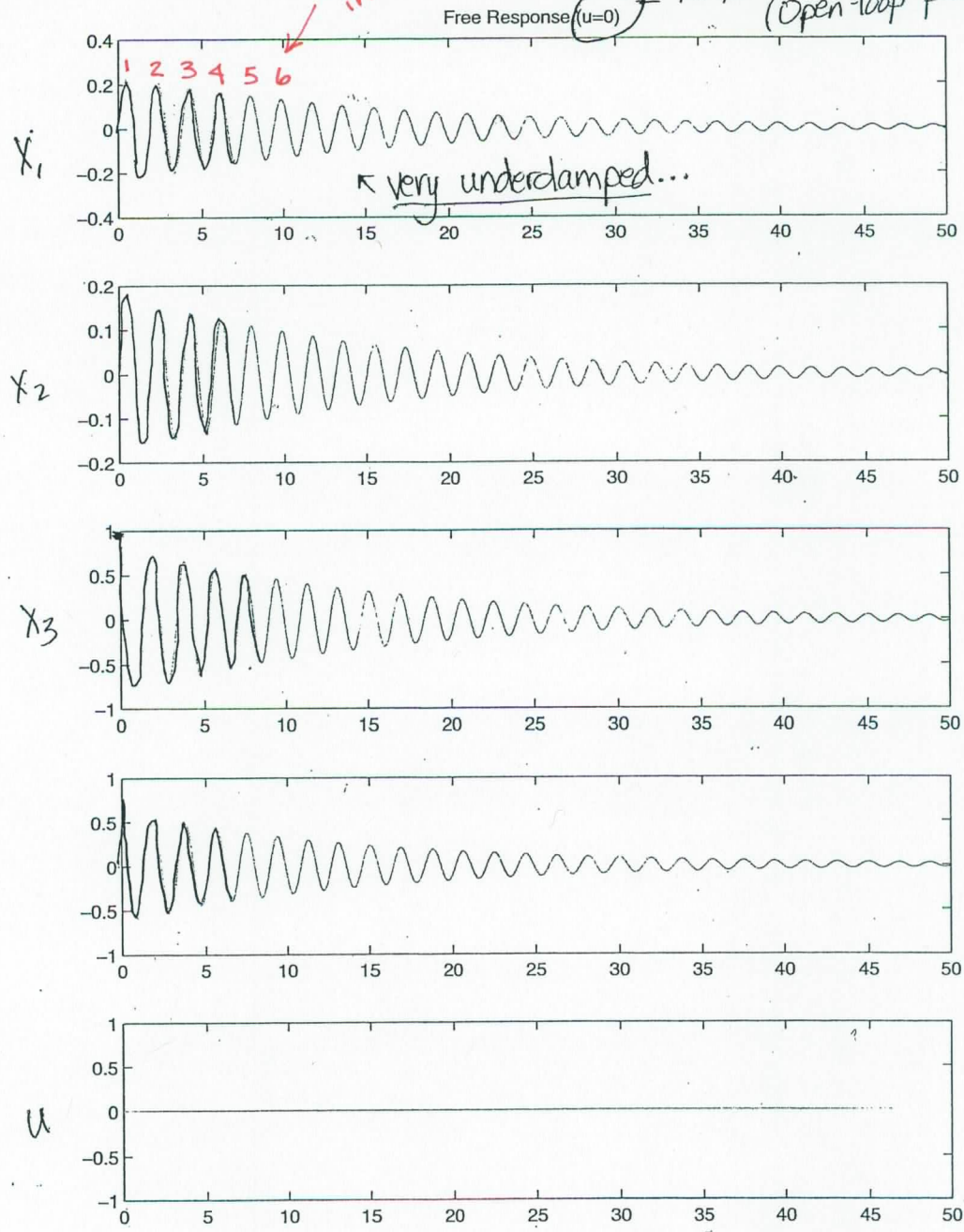
$$K_2 = [0.026, -0.027, 0.046, 0.013]$$

```
69
70 C2: Q1:1.000 Q2:0.000 Q3:0.000 Q4:0.000 R:1.000
71 C2:  0.224  0.118  2.515  1.352  0.008
72 J2 =  0.224 + 0.000 + 0.000 + 0.000 + 0.008 = 0.23157950
73 pdes =
74  -20  -30  -100  -200
```

No control. (Open-loop)

about 6 cycles
in 10 sec, open-loop.

No feedback,
(Open-loop poles)



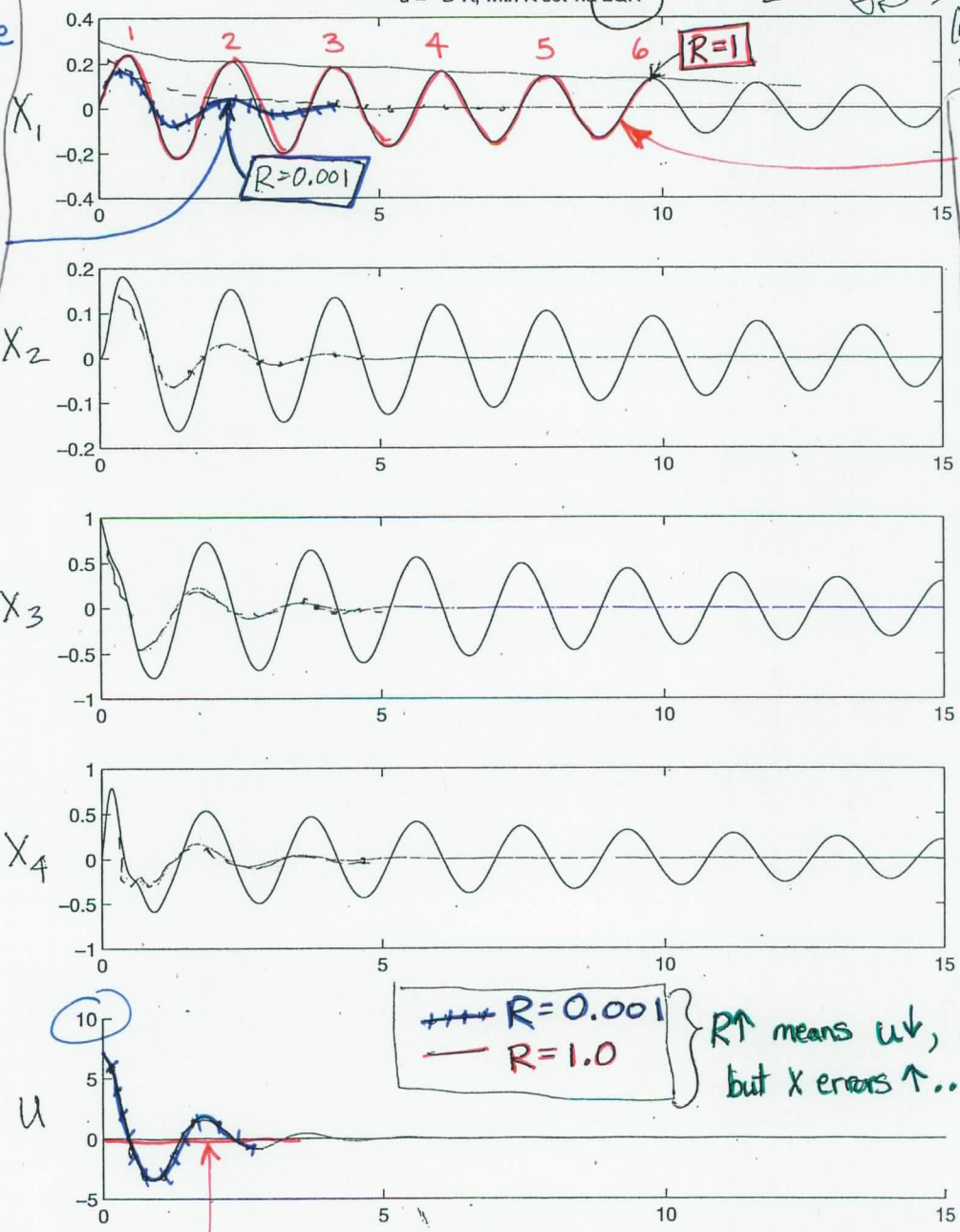
LQR control, to determine K (the gain matrix).

More typically, all diagonal elements of Q have some non-zero value...
 $u = -B^*K$, with K set via LQR

$$Q = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

... & Q is often diagonal. (as a matrix).

w/ $R = .001$, ω_d is close to OL system, but ξ (damping ratio) is much larger...



w/ larger penalty, $R=1$, K is smaller & response of CL system is close to the OL sys. (CL: closed-loop) (OL: open-loop)

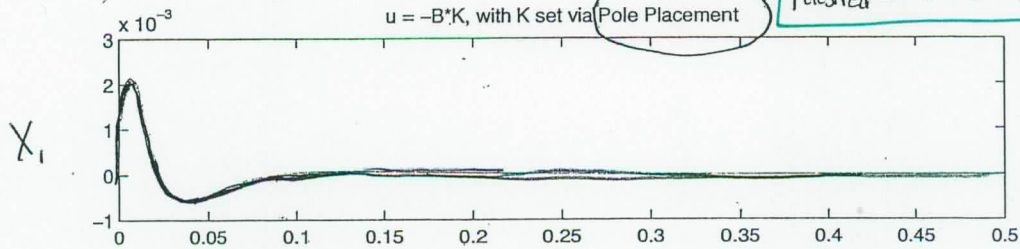
$R \uparrow$ means $u \downarrow$, but x errors $\uparrow$...

For $R=1$: u not exactly "zero", just much smaller in magnitude than when $R=.001$...

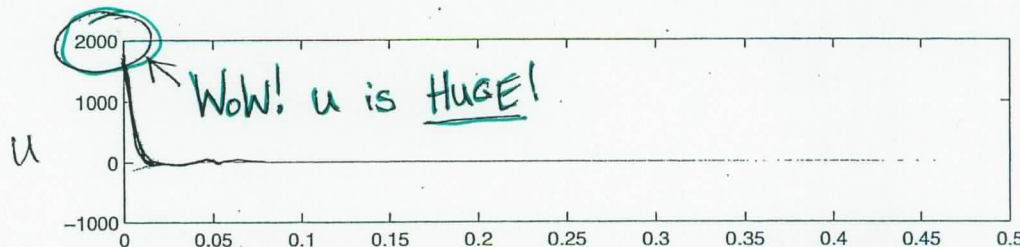
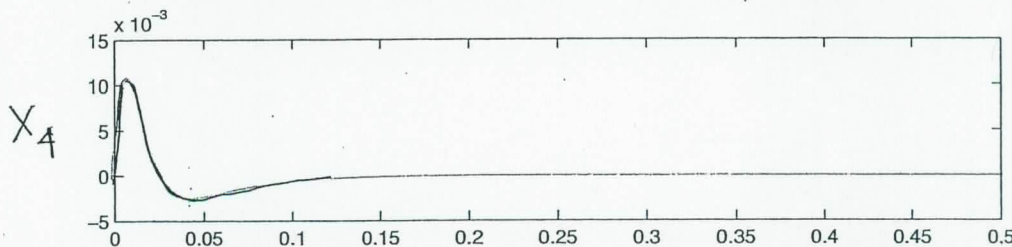
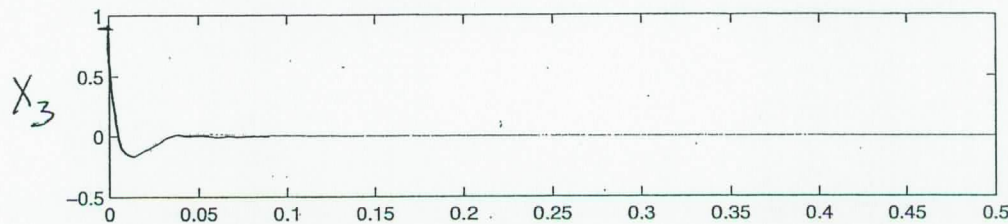
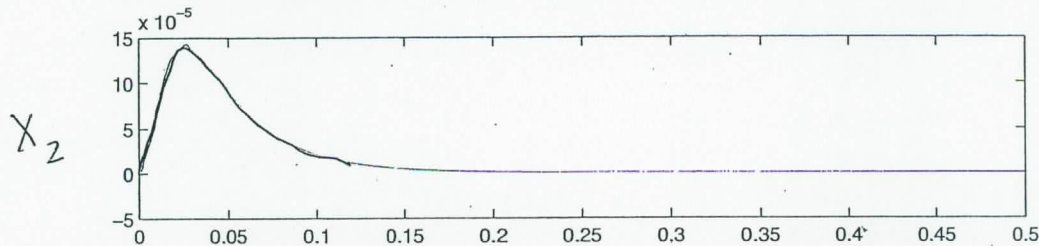
Pole placement, to determine K (gains).

(J is no longer minimized, in any way!)

$$P_{\text{desired}} = [-20, -30, -100, -200]$$



Poles are set rather arbitrarily here...



Wow! u is HUGE!