

Inverse Dynamics : "Inner Loop /
Outer Loop" Approach
(Spong 8.2) ← Craig 10.4

Lecture 16

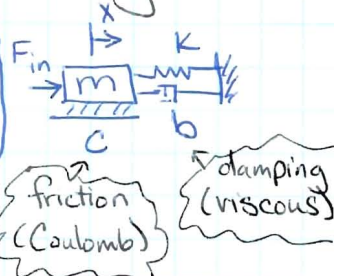
Robot Manipulators

- GEOMETRY affects actuator ↔ end effector ^{dynamic}
- We can use an INNER LOOP to "cancel out" the non-linear robot dynamics

Review!

→ Recall from Lecture 5, "control law partitioning" ...
e.g., Say the EOM (eqn of motion) of a SISO (single-input single-output) system is

$$m\ddot{x} = F_{in} - c \cdot \text{sign}(\dot{x}) - b\dot{x} - Kx$$



- To CONTROL F_{in} , break into two parts:

$$F_{in} = F_{cancel} + F_{control}$$

Where

$$F_{cancel} = c \cdot \text{sign}(\dot{x}) + b\dot{x} + Kx$$

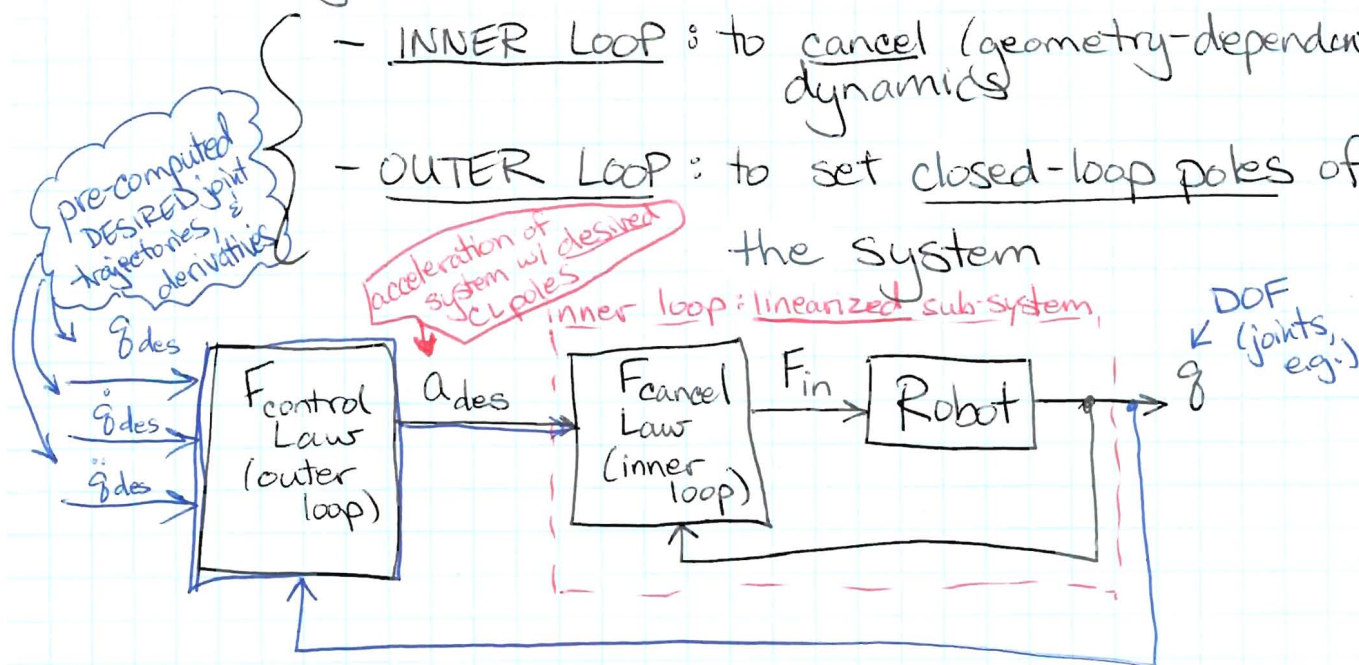
"cancel out" RHS [righthand side]

leaving:

$$m\ddot{x} = (F_{cancel} + F_{control}) - c \cdot \text{sign}(\dot{x}) - b\dot{x} - Kx$$

$$m\ddot{x} = F_{control}$$

- This idea can be generalized for the multi-input multi-output (MIMO) case of n actuators controlling n manipulator joints.
- Visually the architecture has:
 - INNER LOOP: to cancel (geometry-dependent) dynamics
 - OUTER LOOP: to set closed-loop poles of the system



Recall: Jacobian (function of GEOMETRY)

$$\begin{aligned} \dot{\xi}_e &= J \dot{q}_a \\ \tau_a &= J^T F_e \end{aligned} \quad \left\{ \begin{array}{l} a: \text{actuator} \\ e: \text{end effector} \end{array} \right.$$

- EOM (e.g., obtained via the LAGRANGIAN approach) can be written in matrix form as:

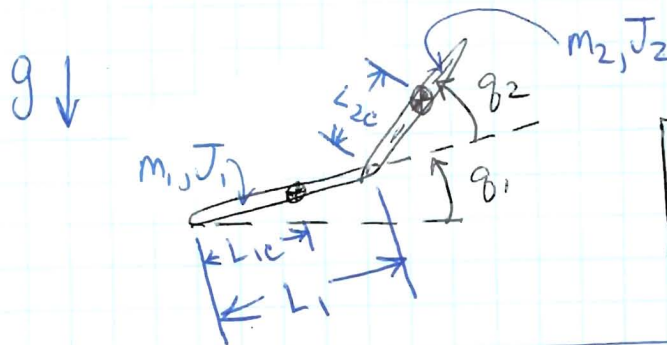
* Manipulator Eqs of Motion

$$M(q)\ddot{q} + C(q,\dot{q})\dot{q} + G(q) + F(q,\dot{q}) = u$$

Labels for the EOM terms:

- $M(q)$: Inertia (mass) matrix
- $C(q,\dot{q})\dot{q}$: Coriolis & centrifugal forces
- $G(q)$: Gravity
- $F(q,\dot{q})$: Friction

Example: Our familiar two-link arm.



Let:

$$h \equiv +m_2 L_1 L_{2c} \sin \theta_2$$

EOM #1

$$\begin{aligned} M_{11} \ddot{\theta}_1 &\rightarrow (m_1 L_{1c}^2 + J_1 + J_2 + m_2 (L_1^2 + L_{2c}^2 + 2L_1 L_{2c} \cos \theta_2)) \ddot{\theta}_1 + \dots \\ M_{12} \ddot{\theta}_2 &\rightarrow (m_2 (L_{2c}^2 + L_1 L_{2c} \cos \theta_2) + J_2) \ddot{\theta}_2 + \dots \\ G_1 &\rightarrow (m_1 L_{1c} \cos \theta_1 + m_2 L_1 \cos \theta_1 + m_2 L_{2c} \cos(\theta_1 + \theta_2)) g + \dots \\ C_1 \dot{\theta} &\rightarrow (-2h \dot{\theta}_1 \dot{\theta}_2 - h \dot{\theta}_2^2) = \tau_1 \end{aligned}$$

first actuator input, to be set via CONTROL

EOM #2

$$\begin{aligned} M_{21} \ddot{\theta}_1 &\rightarrow (J_2 + m_2 L_{2c}^2 + m_2 L_1 L_{2c} \cos \theta_2) \ddot{\theta}_1 + \dots \\ M_{22} \ddot{\theta}_2 &\rightarrow (J_2 + m_2 L_{2c}^2) \ddot{\theta}_2 + \dots \\ G_2 &\rightarrow (m_2 L_{2c} \cos(\theta_1 + \theta_2)) g + \dots \\ C_2 \dot{\theta} &\rightarrow h \dot{\theta}_1^2 = \tau_2 \end{aligned}$$

second input (to be set)

In MATRIX form:

$$\begin{aligned} (M_{21} = M_{12}) \left\{ \begin{bmatrix} M_{11} & M_{12} \\ M_{12} & M_{22} \end{bmatrix} \begin{bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \end{bmatrix} + \begin{bmatrix} -h \dot{\theta}_2 & -h(\dot{\theta}_1 + \dot{\theta}_2) \\ h \dot{\theta}_1 & 0 \end{bmatrix} \begin{bmatrix} \dot{\theta}_1 \\ \dot{\theta}_2 \end{bmatrix} + \begin{bmatrix} G_1 \\ G_2 \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix} \right. \\ \left. \begin{bmatrix} M & \ddot{q} + C \dot{q} + G = \tau \end{bmatrix} \right. \quad (3) \end{aligned}$$

- Note:
- ① M is symmetric ($M_{12} = M_{21}$).
 - ② M is invertible! $\ddot{q} = M^{-1}(u - C\dot{q} - G)$ $\leftarrow$ EOM
 - ③ Equations depend on geometry (non-linear).
 - ④ Equations tend to be "messy"!

Useful to AUTOMATE generation of the EOM from the Lagrangian (e.g., w/ MATLAB).

(Future lecture topic)

Inner/Outer Loop Process.

[1] Inner loop - Assume input a_{des} is the "desired joint accelerations", $\ddot{q}$.

- Then, if $u = M a_{des} + C\dot{q} + G (+ F)$ $\leftarrow$ if friction included

by inspection, since $u = M\ddot{q} + C\dot{q} + G + F$,

the solution will be $\ddot{q} = a_{des}$ given this u .

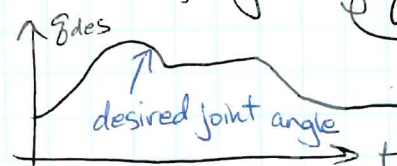
just "cancel" dynamics

[2] Outer loop $\rightarrow$ How to set a_{des} ?

a) Assume we start w/ a desired motion, trajectory q_{des} .

Calculate $\dot{q}_{des} = \frac{d}{dt} q_{des}$

$\ddot{q}_{des} = \frac{d}{dt} \dot{q}_{des}$



④

(2) Outer loop, cont...

b) Use "PD" control on errors in position & velocity:

Feedback Laws:

Note, if $\dot{q} = \dot{q}_{des}$,
Then $\ddot{q}_{des} = \ddot{q}$

$$a_{des} = \ddot{q}_{des} + \left\{ K_p (q_{des} - q) + K_d (\dot{q}_{des} - \dot{q}) \right\}$$

desired actual (measured, or estimated)

PD feedback for errors...

* Recall 2nd-order char. eqn:

$$s^2 + \underbrace{2\zeta\omega_n}_{K_d} s + \underbrace{\omega_n^2}_{K_p} = 0$$

a) Since inner loop ensures $\ddot{q} = a_{des}$ (if NO MODEL ERROR),

$$\ddot{q} = a_{des} = \ddot{q}_{des} + K_p (q_{des} - q) + K_d (\dot{q}_{des} - \dot{q})$$

b) outer loop PD control

Let error be defined as $e = q_{des} - q$

...Then $\dot{e} = \dot{q}_{des} - \dot{q}$

$\ddot{e} = \ddot{q}_{des} - \ddot{q}$

Then:

$$0 = \ddot{e} + K_d \dot{e} + K_p e$$

often, $\ddot{q}$ is used for error, instead of e

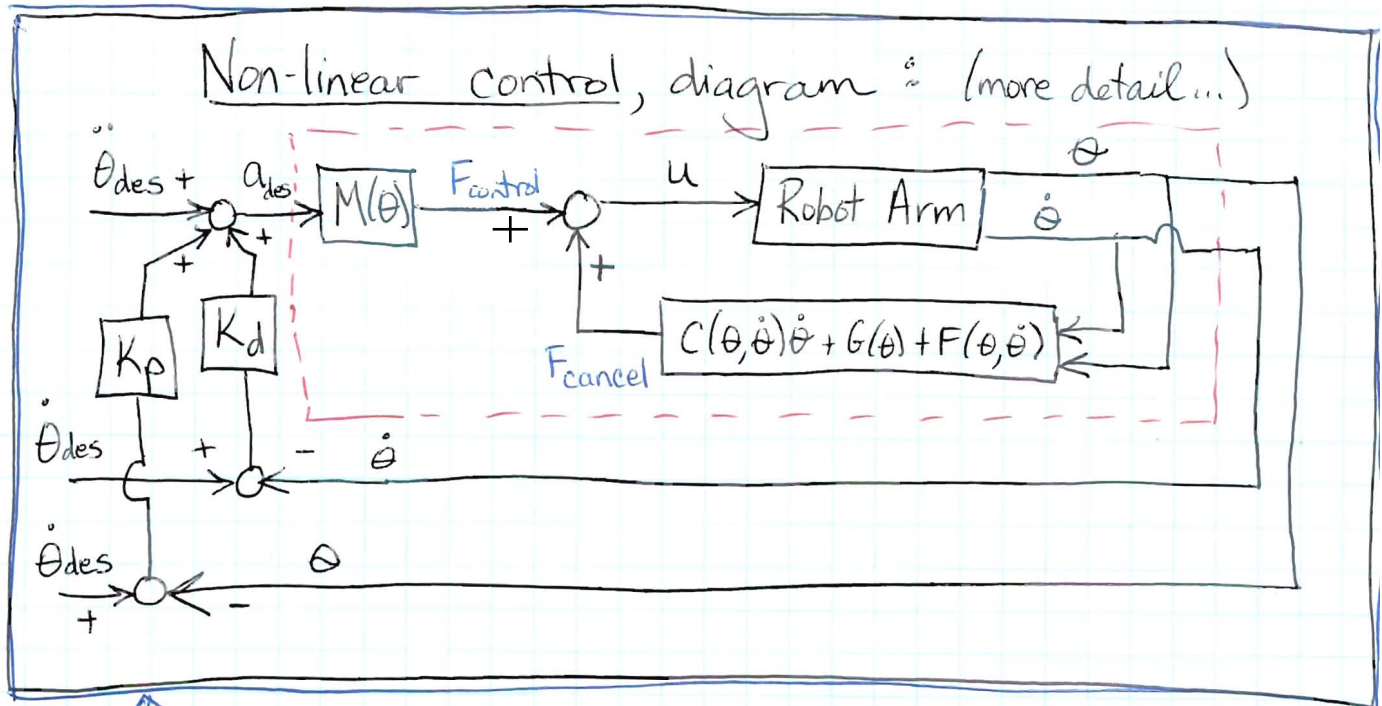
Pick $\zeta \approx \omega_n$ for each joint, e.g., $\zeta = 1$ (or $\zeta = 0.7$)

In general,
 K_p & K_d are
 $n \times n$ diagonal!
n=2 for 2-link arm

$$K_p = \begin{bmatrix} \omega_1^2 & 0 \\ 0 & \omega_2^2 \end{bmatrix}, K_d = \begin{bmatrix} 2\zeta\omega_1 & 0 \\ 0 & 2\zeta\omega_2 \end{bmatrix}$$

(n x n) (n x n)

Recall, our control law is $u = M\ddot{a}_{des} + C\dot{q} + G + F$
(inner loop)



Can use " q " or " θ " here. Using θ as a reminder joint vector is usually angles.

Issues:

- (I) Time! Inner loop calculations may be SLOW!
(Imagine DT control with 1kHz loop... 0.001 sec per step. ☹️)
May not be able to do the math quickly enough?

Solutions (perhaps): (A) Assume we pre-compute M, C, G, F for the planned geometries. Then assume small errors so $M \approx M_d, C \approx C_d, G \approx G_d$, etc. over time.

(B) Update non-linear matrices at a "slow" rates, but keep at small (fast servo rate) for u updates (torques).

- (II) Unknown model params. e.g. a disturbance torque T_d (k)